CONTRACT CHANGE ORDER	Page 1 or 56
CONTRACT CHANGE ORDER	Change Requested by: Engineer
CCO: 176 Suppl. No. 0 Contract No. 04 - 0120F4 Road SF-80-13.2/	13.9 FED. AID LOC. :
To: AMERICAN BRIDGE/FLUOR ENTERPRISES INC A JOINT VENTURE You are directed to make the following changes from the plans and specifications or do the following describe for this contract. NOTE: This change order is not effective until approved by the Engineer	
Description of work to be done, estimate of quantities and prices to be paid. (Segregate between additional vaccount.) Unless otherwise stated, rates for rental of equipment cover only such time as equipment is actual. This last percentage shown is the net accumulated increase or decrease from the original quantity in the Eng	y used and no allowance will be made for idle time.
Adjustment of Compensation at Lump Sum:	
This CCO accepts as "fit for purpose" the localized misalignments and welding transi field splice joints of the Orthotropic Box Girders (OBG) steel deck skin plates as depi "Behavior of Complete Joint Penetration Field Welds Between Orthotropic Box Girde The attached report documents the conditions found in the field, and serves as the atthe alignment issues and the necessary work to mitigate the local misalignment of the OBG.	cted in the attached technical report – r Segments," dated October 28, 2010. ccepted technical document addressing
Adjustment of Compensation at Lump Sum	\$0.00
Estimated Cost: Increase	Decrease \$0.00
By reason of this order the time of completion will be adjusted as follows: 0 Days	
Submitted by Signature Resident Engineer	
X ammy Gull Kannu Balan, S	enior T.E. Date Z-1-1/
Approval Recommended by Signature Supervising Bridge Engineer	
TO Uffel Mod P-William Casey,	Sup. B.E. Date 2 - / - //
Engineer Approval)by Signature Principal Transportation Engineer	
Peter Siegenthat We the undersigned contractor, have given careful consideration to the change proposed and agree, if this equipment, furnish the materials, except as otherwise be noted above, and perform all services necessary payment therefor the prices shown above.	s proposal is approved, that we will provide all
NOTE: If you, the contractor, do not sign acceptance of this order, your attention is directed to the proceeding with the ordered work and filing a written protest within the time therein specified.	requirements of the specifications as to
Contractor Acceptance by	
Signature (Print name and title) BRIAN A. PETERSEN - PROJE	CTDRECTOR DAFEBII

DL McQUAID & ASSOCIATES Inc.

2306 Cassidy Drive Bethel Park, PA 15102 Phone No. 412-831-9177 Fax, No. 412-851-4116

To: American Bridge / Fluor Enterprises Inc., JV

October 22, 2010

Mr. Mike Flowers Project Director 375 Burma Road Oakland, CA 94607

Phone No.

510-808-4600

Fax No.

510-808-4601

Re: / Review of Expert Report by John M. Barsom and Kevin M. Smith – Behavior of Complete Joint Penetration Field Welds Between Orthotropic Box Girder Segments

Dear Mr. Flowers:

I have reviewed the Technical Report – Behavior of Complete Joint Penetration Field Welds Between Orthotropic Box Girder Segments by John M. Barsom and Kevin M. Smith as requested and I offer the following letter in support of the technical content and accuracy of the report. Based on this review I recommend that the report be used by all parties to accept the misalignment of the orthotropic box girders.

I provided Mr. Barsom information as he required in the development of the expert report and have discussed with him the techniques used to fabricate, erect and weld the orthotropic box girders at the ZPMC facility in China and at the project site in Oakland California. I also visited the project site and witnessed the fit-up and welding and can verify that the field welding was done as described in the expert report and is in general conformance with the requirements of the AWS D1.5 Bridge Welding Code and the workmanship standards expected of a modern day steel erector. I have reviewed the stress calculations and they appear to be acceptable, however, that is not my area of expertise and I will not comment on them in detail. The expert report has addressed the alignment issues in depth and has provided the technical information to justify accepting the orthotropic box girders for the SAS Bridge.

Based on my discussions with American Bridge/Fluor, Mr. Barson and my involvement as a welding consultant to the Joint Venture it is my professional opinion based on a reasonable degree of engineering certainty that the expert report accurately describes the condition of the fit-up and welding at the project site, is technically accurate and I recommend that it be accepted as written and submitted to Caltrans for their approval.

Sincerely

D. L. McQuaid P.E.

Welding Consultant

BEHAVIOR OF COMPLETE JOINT PENETRATION FIELD WELDS BETWEEN ORTHOTROPIC BOX GIRDER SEGMENTS

SAN FRANCISCO-OAKLAND BAY BRIDGE

SELF-ANCHORED SUSPENSION SPAN

BY

John M. Barsom

Barsom Consulting Ltd

u/ic/zoi

And

Kevin M. Smith

American Bridge / Fluor Enterprises Inc., A Joint Venture

INTRODUCTION

Because it is a "self-anchored" suspension bridge most of the orthotropic box girder deck structure will be under significant axial compression when completed. This compression eliminates, or minimizes, concern regarding the fatigue performance of the transverse field deck splices that are under compression. This is evidenced by the Contract Document provision that permits transverse backing bars used for the complete joint penetration (CJP) field welds to remain in place for all but the two most eastern deck splices – those between Lifts 12 and 13 and those between Lifts 13 and 14. Bridges with orthotropic deck plates commonly have the transverse backing removed to minimize potential fatigue damage. Compression in the eastern portion of the SFOBB structure is reduced as a result of shear lag and dead and live load flexure stresses.

The American Welding Society (AWS) D1.5-2002 Bridge Welding Code is the governing document for the welded field splices. It states in Section 3.3.3: "Where parts are effectively restrained against bending due to eccentricity in alignment, the offset from theoretical alignment shall not exceed 10 percent of the thickness of the thinner part joined, but in no case shall be more than 3mm [1/8 in.]." Furthermore, it states in Section 3.3.1.1: "The separation between faying surfaces of plug and slot welds, and butt joints landing on a backing, shall not exceed 2 mm [1/16 in.]."

The following discusses the fatigue implications of deck plate misalignments at the field splices greater than those allowed by AWS D1.5, and addresses backing bar separations that are more than 2mm.

Page 1 of 6

October 28, 2010

DISCUSSION

Dimensional tolerances specified for the fabrication of the steel box girders allow variations in the overall depth, skin plate flatness, and the placement and alignment of stiffeners and other components between adjacent segments. As an example, the specified tolerance for the depth of box girder segments is +8mm/-5mm. In order to achieve a "best fit" between segments significant effort has been expended during both trial and in-field fit-up to accommodate the dimensional variations. For an overview of the welding and bolting operation used to align and connect the Orthotropic Box Girder top deck plates in the field, reference Appendix B. Notwithstanding the efforts to align the field splices, offsets greater than those allowed by AWS D1.5 have occurred over localized areas of the deck structure.

For the transverse deck plate field splices completed to date for OBG Lifts 1E through 7E and 1W through 6W, the plates are typically aligned within AWS D1.5 tolerances of 2mm; however, at a few locations the offsets exceed the 2mm alignment tolerance by up to 4mm. Reference the offset maps provide in Appendix C. The average offset between adjacent deck plates is 0.33mm with a standard deviation of 1.15mm. The accumulated out-of-tolerance deck misalignments along a box girder splice are, on average, not more than 7% of the deck width, with a peak offset of 6mm. The average length of the out-of-tolerance sections is less than 400mm. Because the deck plate is highly restrained at the north and south edges of the box girder where the deck forms a corner with the adjoining edge plate, there is a higher frequency of out-of-tolerance deck misalignments at these locations. These misalignments are located out of the traveled way where fatigue from direct wheel loads is not a concern and where the thickness of the deck plate is 20mm, contrary to the majority of the deck comprising the roadway shoulders and traveled way which is 14mm thick, except at the far west and east ends where it thickens to 20mm.

In some instances the planar misalignment has prevented a tight fit between the transverse backing bar and the underside of the deck plates causing maximum separations of up to 4mm over short lengths. A procedure has been developed to fill these gaps using a shielded metal arc process prior to implementing flux-cored and submerged arc welding for the completion of the CJP groove welds.

When field welding at a transverse deck splice is complete the weld reinforcement is ground flush. In areas of misalignment grinding has produced sloped transition surfaces that are no steeper than 2.5 to 1. The AASHTO fatigue category for field splices that do not have the backing removed, with or without sloped transitions, and regardless if the crown is ground flush or not, is category "C". This is the fatigue category representing all but the two eastern field splices. If the weld reinforcement, or backing, is removed and the welds are ground to a slope not exceeding 2.5 to 1 the fatigue category is "B". This is the category that will apply to the two most eastern deck splices and will provide the highest fatigue category for a welded member – category "A" being reserved for plain members comprised of base metal with rolled, thermal cut, or cleaned surfaces.

Planar misalignments at the deck splices may result in secondary stresses due to the resultant vertical eccentricity between the adjoining plates. As noted above, the planar misalignments are localized, occurring over short lengths and average no more than 7% of the deck width. The unfactored axial compression stresses from dead load and live load, combined with the secondary

flexural stresses due to a planar misalignment of 6mm have been calculated (reference Appendix D) to be 385MPa in the 14mm thick deck plate. While this stress is larger than the minimum required yield strength of the material, 345MPa, it should be noted that the actual yield strength of the 14mm plate varies between 364MPa and well above 400MPa with an average yield strength of 432MPa. Because of the higher yield strength yielding of the 14mm thick deck plate is not expected. However, because the misalignments are over short lengths and when combined represent only a small percentage of the overall deck width, if yielding were to occur at discrete locations, the resulting strain would result in a redistribution of stress that would mitigate any adverse impacts. Additionally, the reduction in stress due to the presence of the backing bar, the 2.5 to 1 sloped transitions and the 50mm epoxy asphalt concrete overlay have not been considered here but are suspected to reduce the stress in the misaligned portions of the deck plate. Therefore, secondary stresses due to a planar misalignment of 6mm have been determined to be within acceptable limits for service loads.

Fatigue Behavior at the Weld Root with Backing - Deck in Compression

Typically, solidification of deposited weld metal induces residual compressive stresses at the root of single-V CJP welds, such as those for the deck splices. Fatigue cracks rarely initiate under compression-compression cyclic loading. The size of these cracks, should they occur, are limited to the size of the plastic zone at the location of any stress riser. The crack size will be limited to a few thousands of an inch, a value less than what the code allows for discontinuities in base and weld metals that are inspected by ultrasonic testing. Furthermore, because the transverse splices are under significant axial compression, if cracks were to initiate they would not propagate under cyclic compressive stresses. This case applies to all but the two most eastern deck field splices.

Fatigue cracks, if they were to initiate between the deck plate and the backing bar, would initiate at the corner between the bottom surface of the deck plate and the root of the deposited weld metal. The attached figure in Appendix A presents a typical cross section of deck plates, backing bar, shielded metal arc welds with a nominal gap between the bottom surface of the deck plate and the top surface of the backing bar. The dimensions of the 9/16-inch (14mm) thick deck plate, the 1½-inch wide backing bar and the cross section of the welds are proportioned to correctly represent the dimensions of the field welded joint. The figures indicate that a deck plate planar misalignment may produce a gap between the bottom surface of the deck plate and the top surface of the backing bar of up to 1/8-inch. The geometry at the corner between the bottom surface of the deck plate and the deposited weld bead does not change significantly when the gap varies from zero through 1/8-inch. Consequently, the stress concentration at this location is negligibly affected by the gap between steel backing and the deck plate. Therefore, the performance of the groove weld is not expected to be different for gaps ranging from 0mm to 4mm.

AASHTO LRFD Section 6.6.1.2.1² states that residual stresses shall not be considered in investigating fatigue and that the provisions for fatigue shall be applied only to details subjected to a net applied tensile stress. In regions where the unfactored permanent loads produce compression, fatigue shall be considered only if the compressive stress is less than twice the maximum tensile live load stress resulting from the fatigue load combination specified in Table 3.4.1-1. As demonstrated in the calculations provided in Appendix D, the dead load compressive

stresses for splices 1 through 10 range from 83MPa to 104MPa. These dead load compressive stresses are greater than twice the tensile live load stress of 40.5MPa. For splice 11 the dead load compressive stress is 72MPa. It is expected that consideration of the wheel load distribution due to the epoxy asphalt concrete overlay will reduce the tensile live load stress from 40.5MPa to less than 34MPa. Under these conditions, load-induced fatigue need not be considered for the deck welds in compression-compression cyclic loading.

Fatigue Behavior at the Crown of the Weld - Deck in Compression

Based on geometry alone – neglecting the fact that the deck plate is not subjected to cyclical tension stresses under normal operating conditions – in the absence of rejectable weld metal indications, the fatigue performance of the field splices will be dictated by the transition between the two plates. Fatigue cracks rarely initiate under compression-compression cyclic loading, however, if they were to form, they would initiate at the toe of the weld. Fatigue cracks propagate perpendicular to the direction of the applied cyclic stresses. Because of the inclined side of the single-V groove, fatigue cracks, if they were to occur, would propagate from the toe of the weld into the base metal of the top deck plate. However, the propagation would not proceed because the deck plate is not subjected to tension cyclic stresses under normal operating conditions.

The preceding discussion demonstrates that fatigue cracks should not initiate or propagate from the face of the groove welds between the top deck plates.

CJP groove welds with backing and excess weld reinforcement correspond to AASHTO fatigue category "C". Though not accounted for, grinding the face of the weld to a 2.5 to 1 slope improves the geometry and fatigue resistance at the face of the weld. It should be emphasized that for all but the last two field splices these welds are subjected only to compressive cyclic stresses under normal operating conditions, a condition under which fatigue cracks rarely initiate and do not propagate.

Fatigue Behavior of Welds - Deck without Significant Compression

The box girder field splices at PP117.5 and PP124.5 are subjected to a significant amount of negative dead load moment which reduces the dead load compressive stresses in the deck plate. The deck plate compressive stresses are further reduced at these splices because the box girder cross sections are more robust than the typical box girder cross section. The area and moments of inertia are approximately 40% to 70% more at these cross sections than the typical bridge cross sections. Additionally, the deck plate thickness at these splices is 20mm instead of the 14mm thick deck plate used elsewhere.

At these field splices the backing bar will be removed and the 2.5 to 1 sloped transitions will be present at both the top and bottom surfaces of the deck plate, effectively eliminating any stress risers. The un-factored axial compression stresses from dead load and live load, combined with the secondary flexural stresses due to a planar misalignment of 6mm have been calculated to be 96MPa and 111MPa for the box girder field splices at PP117.5 and PP124.5, respectively. Therefore, secondary stresses at these splices have been determined not to be significant.

The stress range at these splices will be less than that previously calculated for the typical splices due to the reduced span length between floor beams and the increased deck plate thickness. The fatigue category for these splices is "B", which permits a higher Constant-Amplitude fatigue threshold.³ The fatigue stress range at PP117.5 and PP124.5 is 23MPa which is less than the nominal fatigue resistance of 55MPa. Therefore, load induced fatigue including secondary effects is acceptable for the two most eastern deck field splices.

Additional locations to be considered:

The Department has independently measured the planar alignment and has reported a few notable differences from the values provided in Appendix C. A maximum offset of 13mm has been identified at the intersection of the edge plate and the deck plate. This misalignment occurs at a single location over a length not more than 100mm. This offset is located outside the traveled way at the edge of the box girder where traffic live load is not present and fatigue is not a concern. The deck plate at this location is 20mm thick and is highly restrained due to the proximity of the misalignment to the edge plate. At this location the misaligned portion of the deck plate has been transitioned at 2.5:1 slopes on both the top and bottom of the deck plate and live loads are not present; therefore, stresses, including secondary stresses, at this location are not a concern.

The Department also noted a 7mm offset over an 800mm length at the field splice between Lift 1E and 2E and an 8mm offset over a 200mm length at the field splice between Lift 2E and 3E. These offsets are located within the traveled way and are more than the 6mm offset measured by Smith Emery Company, ABFJV's independent welding quality control inspector. At these locations, the un-factored axial compression stresses from dead load and live load, combined with the secondary flexural stresses due to a planar misalignment of 8mm have been calculated to be 259MPa and 387MPa in the 14mm thick deck plate (reference Appendix D). These stresses are equivalent to the maximums reported elsewhere on the bridge and have been determined to be within acceptable limits for service loads. Fatigue loading at these locations only increases slightly and the dead load compressive stresses are still greater than twice the tensile live load stress of 43.1MPa. Therefore, load-induced fatigue need not be considered for these deck welds in compression-compression cyclic loading.

CONCLUSION

The preceding discussion indicates that:

- 1. Compression stress in the deck plate and the compressive residual stress at the root of the single-V CJP weld will inhibit fatigue crack initiation and propagation at any gap between the backing bar and the deck plate.
- 2. Compression stress in the deck plate will prevent fatigue cracks at the face of the sound single-V CJP weld in areas of planar misalignment. If fatigue cracks were to initiate they would not propagate due to the compressive stress.
- 3. Groove-welded deck field splices between Lifts 12 and 13 and Lifts 13 and 14, where the backing is removed, the top and bottom faces ground, and transitions have slopes no steeper than 2.5 to 1, will have no stress risers. Secondary stresses have been determined not to be significant. Under these conditions the field splices will provide acceptable fatigue resistance.
- 4. The performance of the groove-welded deck field splices with backing bar gaps ranging from zero to 2mm, with additional 2mm tolerance, will not differ substantially from one another.

References:

- 1. <u>AASHTO LRFD Bridge Design Specifications, Second Edition, 1998, Table 6.6.1.2.3-1</u> Detail Categories for Load-Induced Fatigue.
- 2. <u>AASHTO LRFD Bridge Design Specifications, Second Edition, 1998, Section 6.6.1.2.1</u> <u>Load-Induced Fatigue, Application.</u>
- 3. <u>AASHTO LRFD Bridge Design Specifications, Second Edition, 1998, Table 6.6.1.2.5-3</u> <u>Constant-Amplitude Fatigue Thresholds.</u>

APPENDIX A

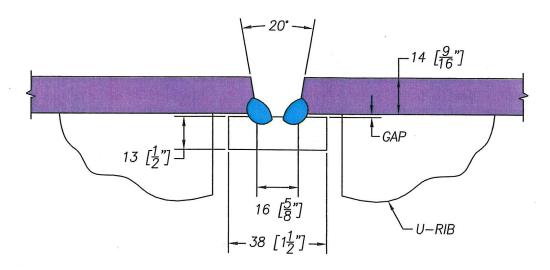


FIGURE 1
TYPICAL DECK JOINT CROSS SECTION (DECK PLATES ALIGNED)

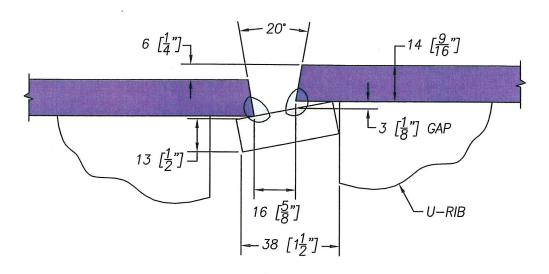


FIGURE 2
TYPICAL DECK JOINT CROSS SECTION (DECK ALIGNED TO 6mm)

SAN FRANCISCO OAKLAND BAY BRIDGE EAST SPAN SEISMIC SAFETY PROJECT SELF ANCHORED SUSPENSION BRIDGE (SUPERSTRUCTURE AND TOWER) STATE OF CALIFORNIA
DEPARTMENT OF TRANSPORTATION
CONTRACT NO. 04-0120F4
BRIDGE NO. 34-0006L/R

DISTRICT COUNTY ROUTE KILOMETER POST 04 SF 80 13.2 / 13.9



Design By: K. SMITH Date: 09/30/10 Design Chk: D. HESTER Date: 09/30/10 Drawn By: K. SMITH Date: 09/30/10 09/30/10 Drawing Chk: D. HESTER Date: In Charge Of: J. CALLAGHAN AB Job No. 660110

ORTHOTROPIC BOX GIRDER DECK JOINT CROSS SECTION

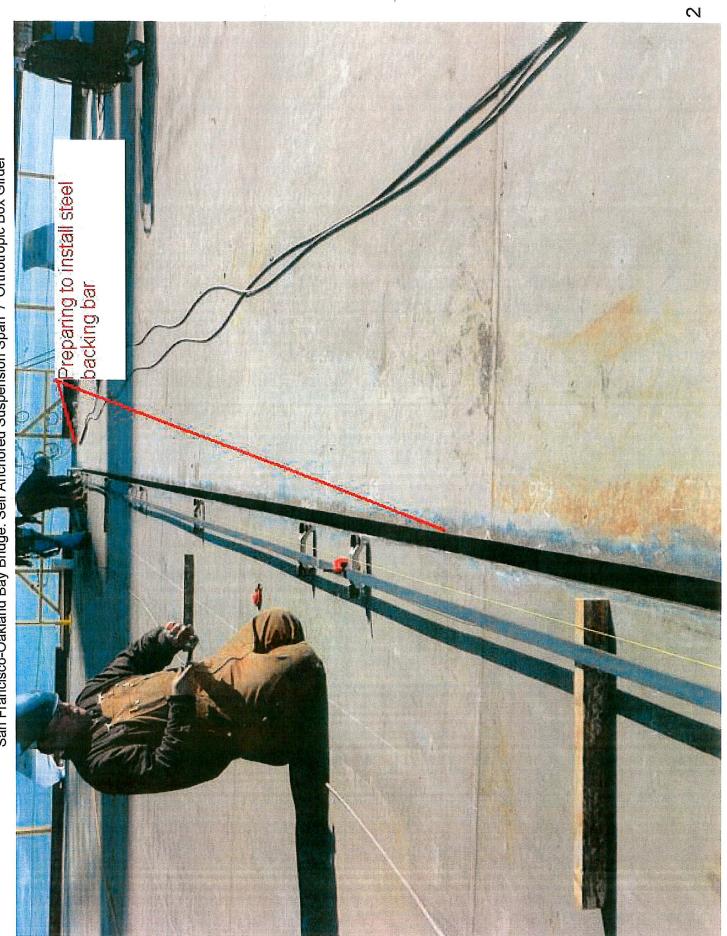
Scale: NTS

Sheet No. 1 of 1

TAT T TO TAKE TO	APP	END	XI	B
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OVERVIEW OF THE BOX GIRDER BOLTING AND WELDING OPERATION

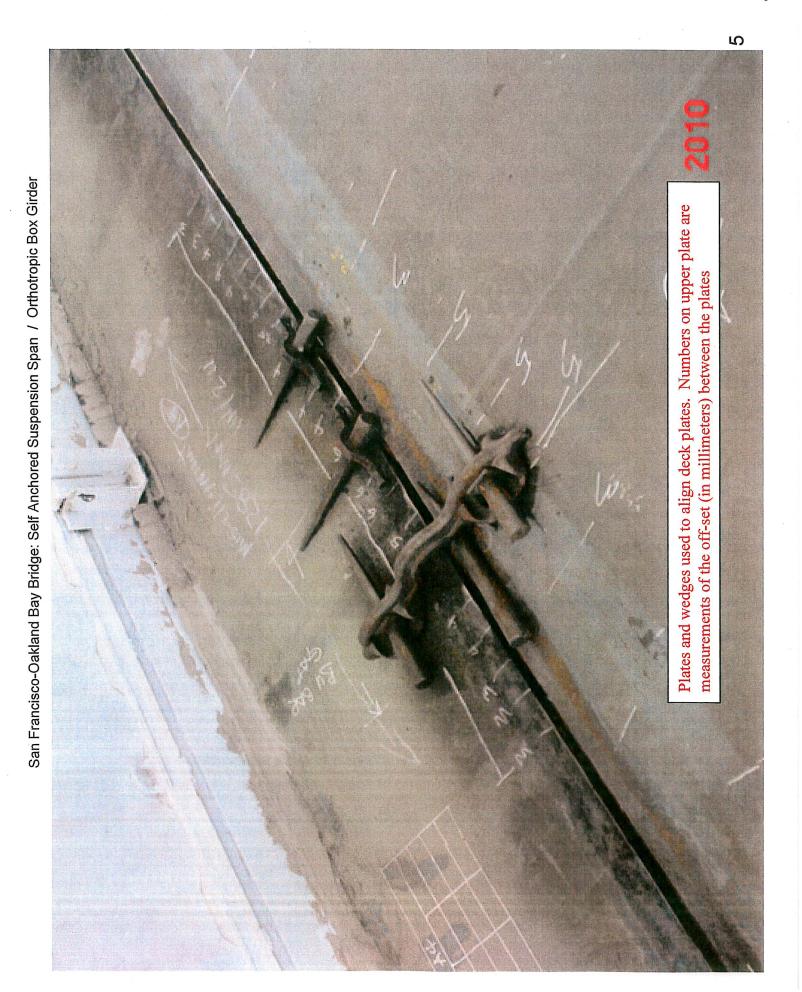
operation for Top Deck Plates for Overview of welding and bolting Lifts 1-6

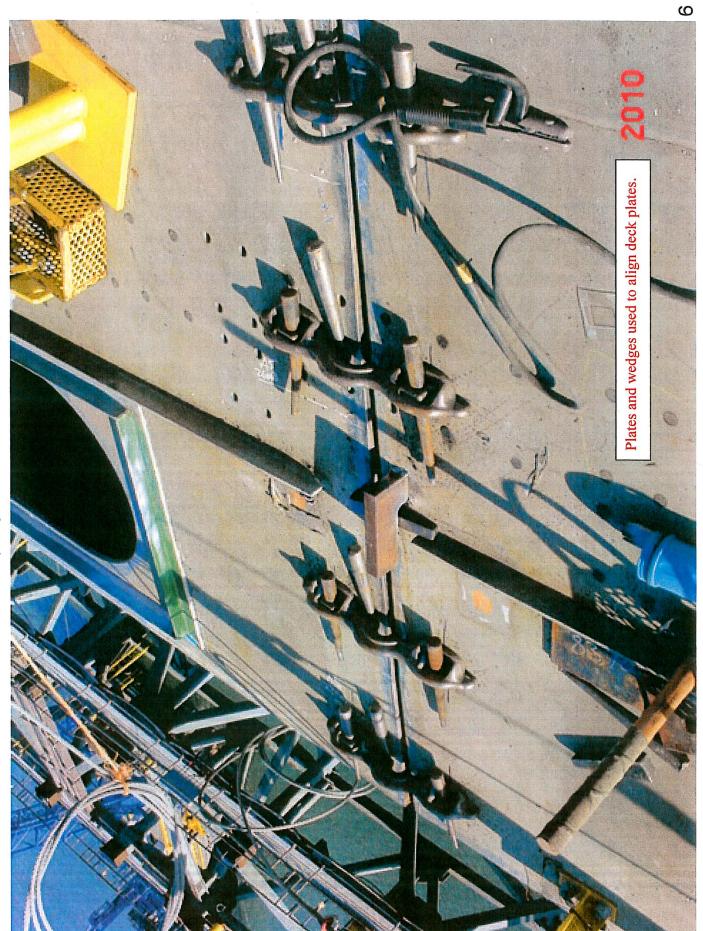




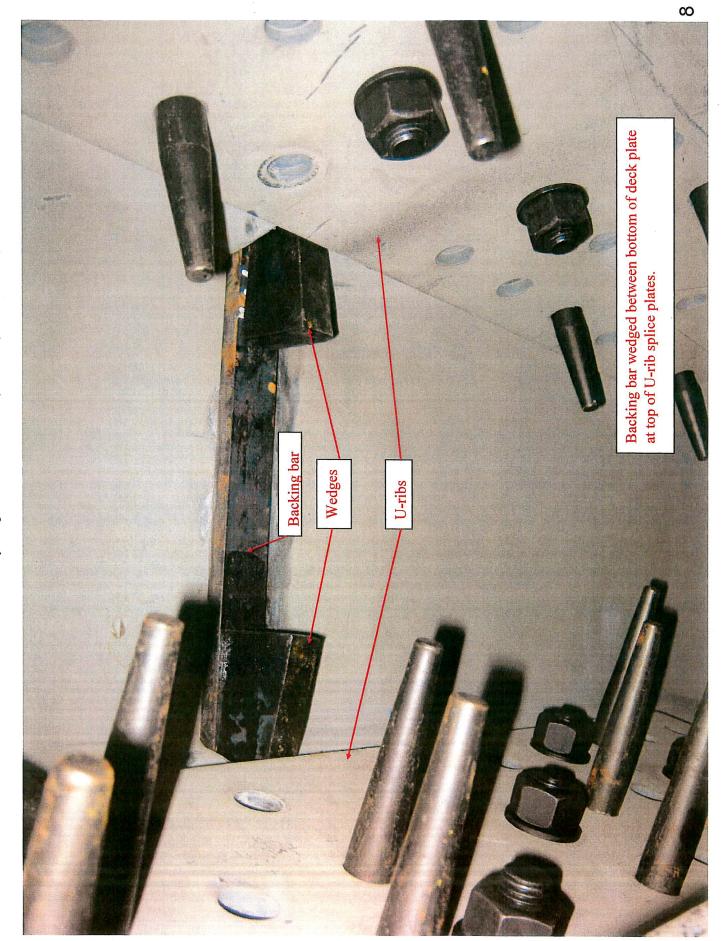


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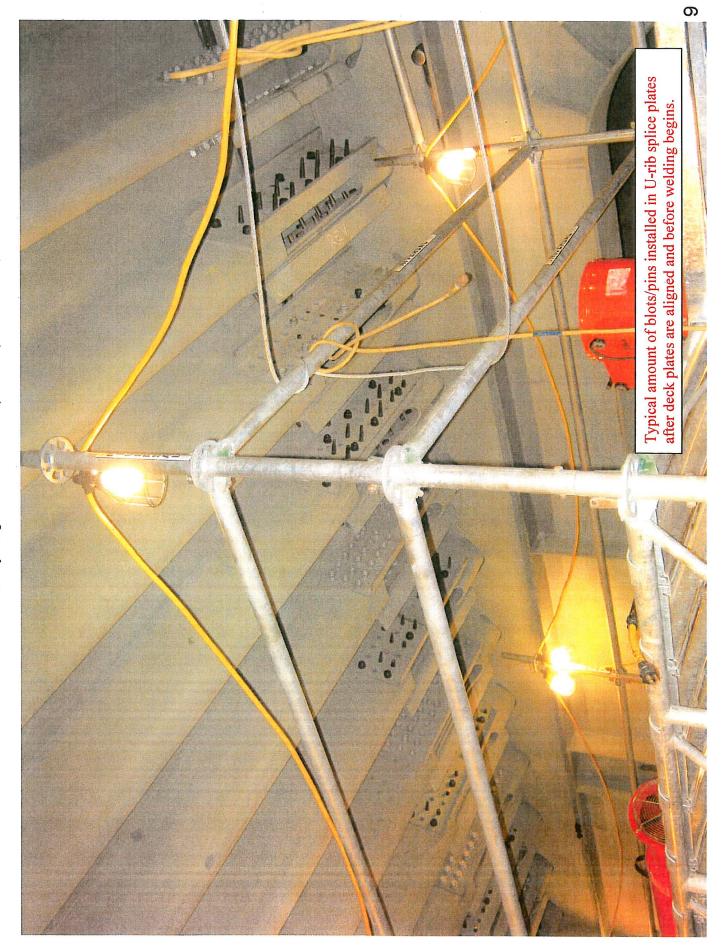




San Francisco-Oakland Bay Bridge: Self Anchored Suspension Span / Orthotropic Box Girder



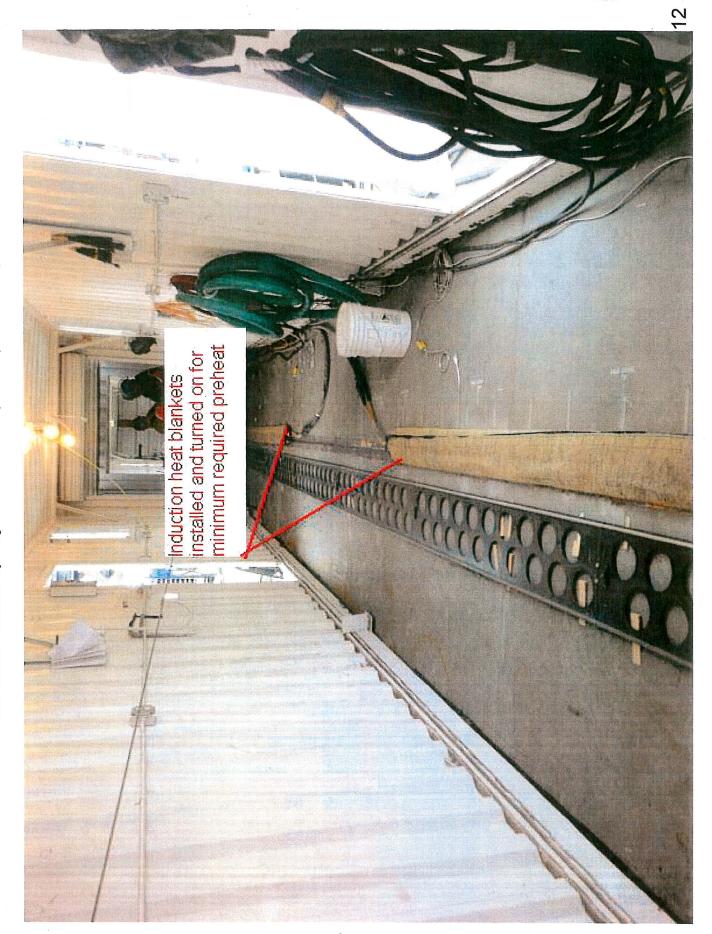
San Francisco-Oakland Bay Bridge: Self Anchored Suspension Span / Orthotropic Box Girder



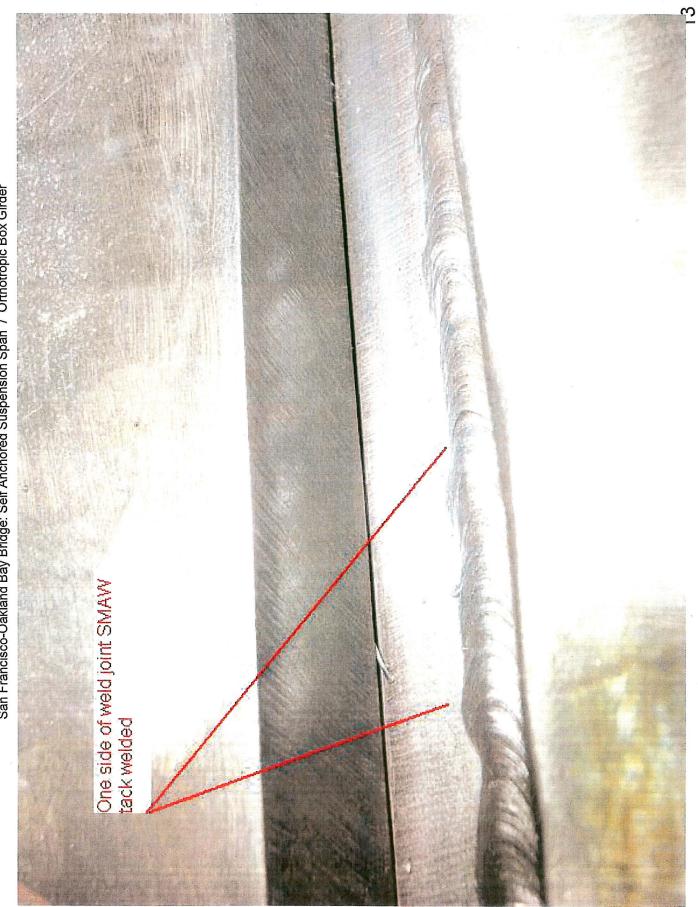
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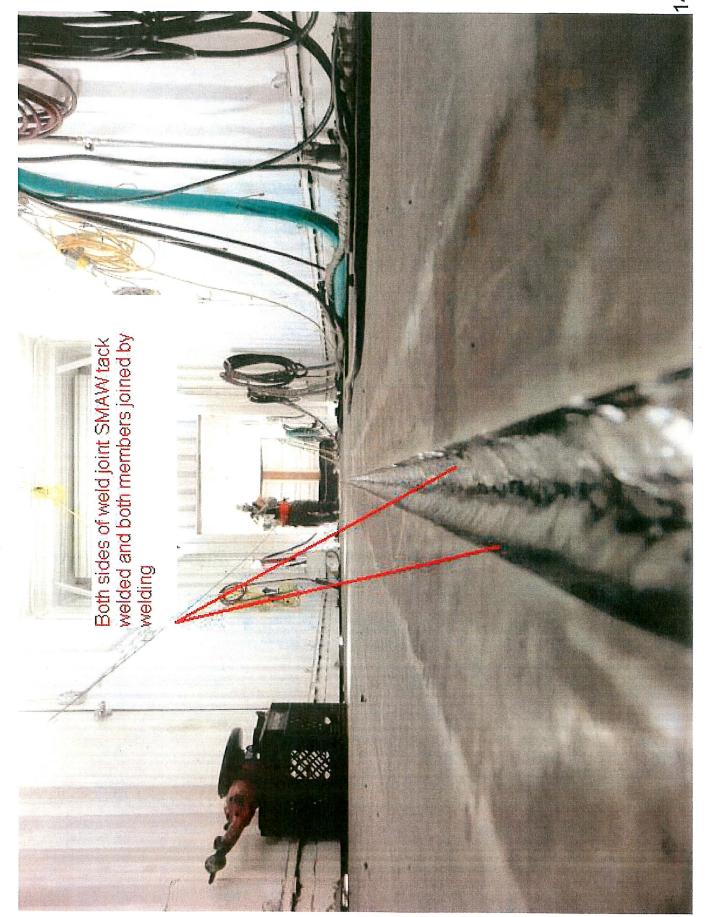




San Francisco-Oakland Bay Bridge: Self Anchored Suspension Span / Orthotropic Box Girder



San Francisco-Oakland Bay Bridge: Self Anchored Suspension Span / Orthotropic Box Girder





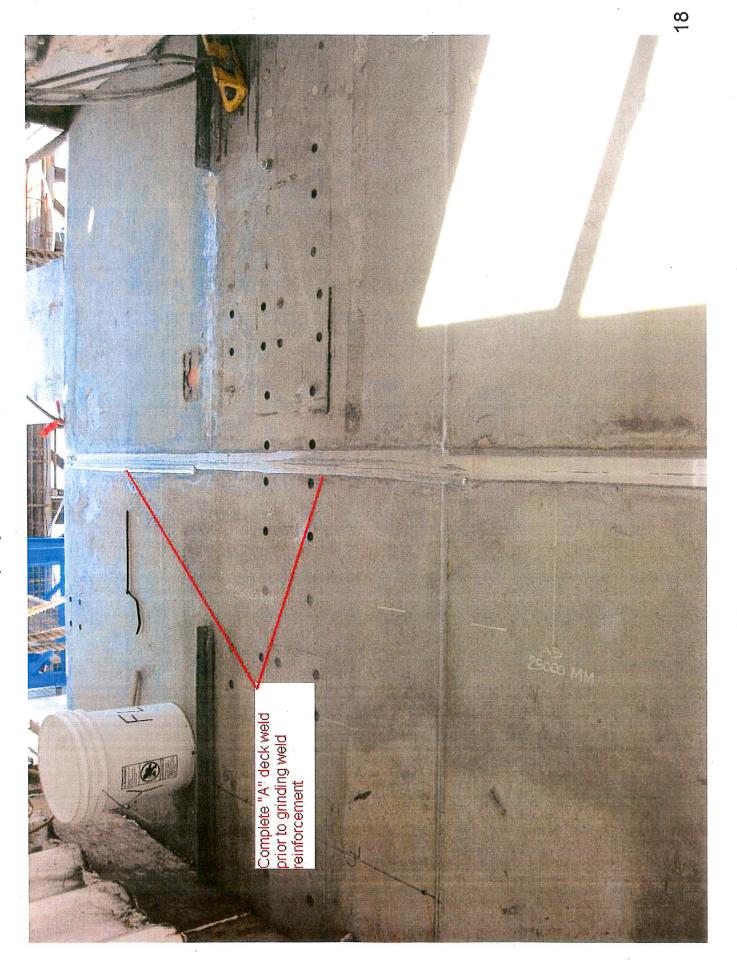
San Francisco-Oakland Bay Bridge: Self Anchored Suspension Span / Orthotropic Box Girder



San Francisco-Oakland Bay Bridge: Self Anchored Suspension Span / Orthotropic Box Girder



San Francisco-Oakland Bay Bridge: Self Anchored Suspension Span / Orthotropic Box Girder



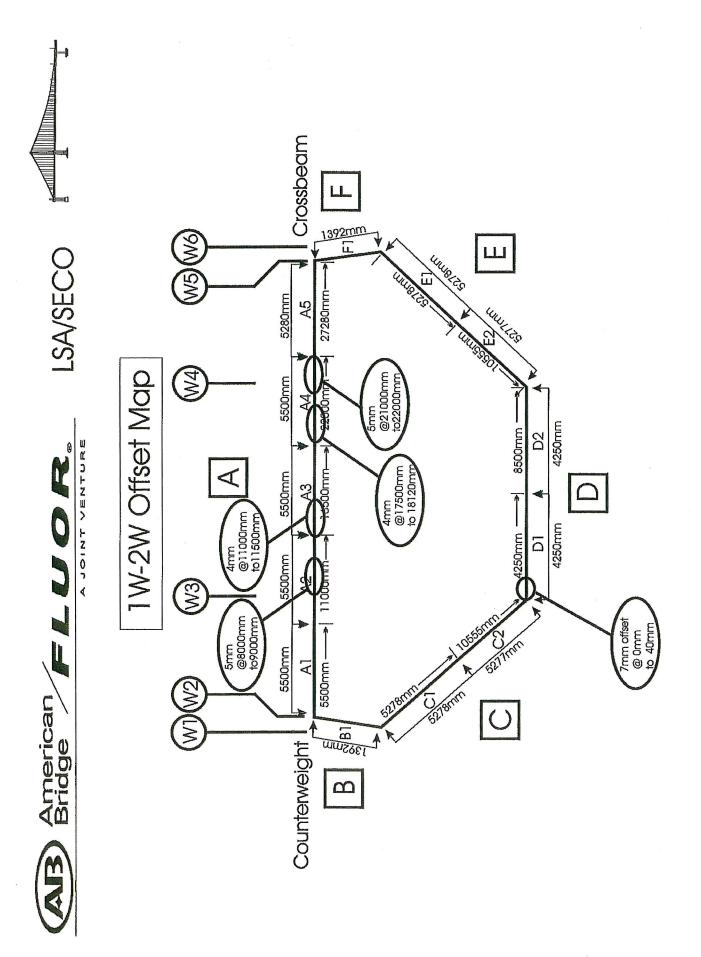
San Francisco-Oakland Bay Bridge: Self Anchored Suspension Span / Orthotropic Box Girder

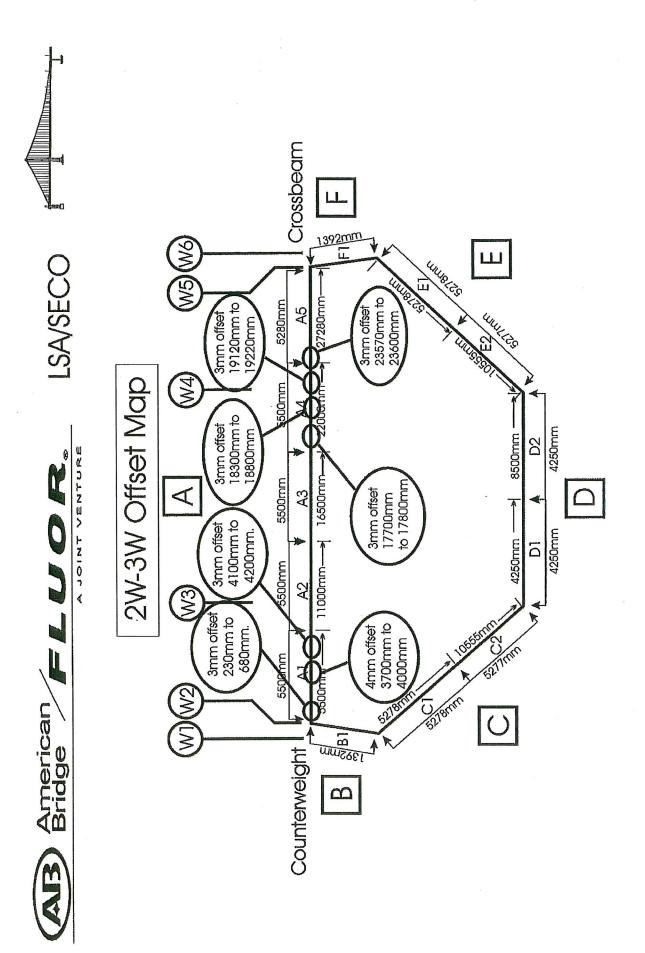


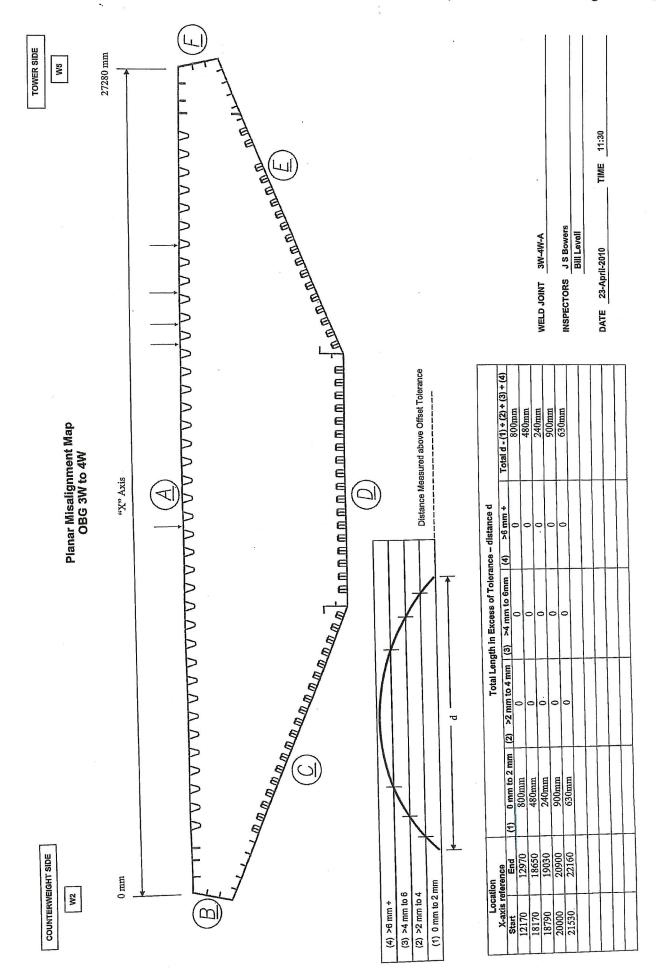
San Francisco-Oakland Bay Bridge: Self Anchored Suspension Span / Orthotropic Box Girder

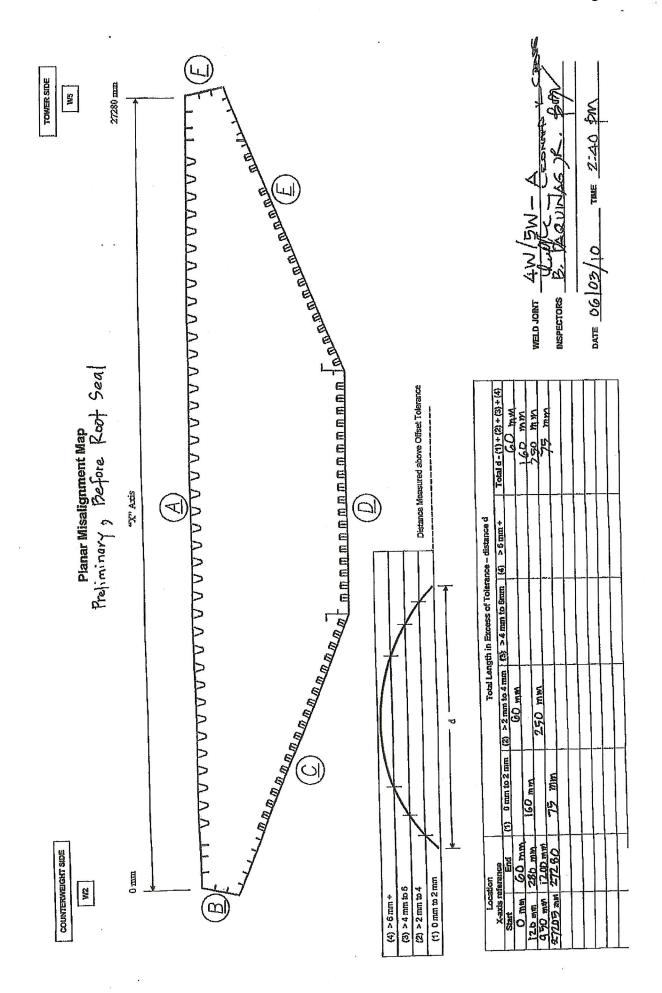
APPENDIX C

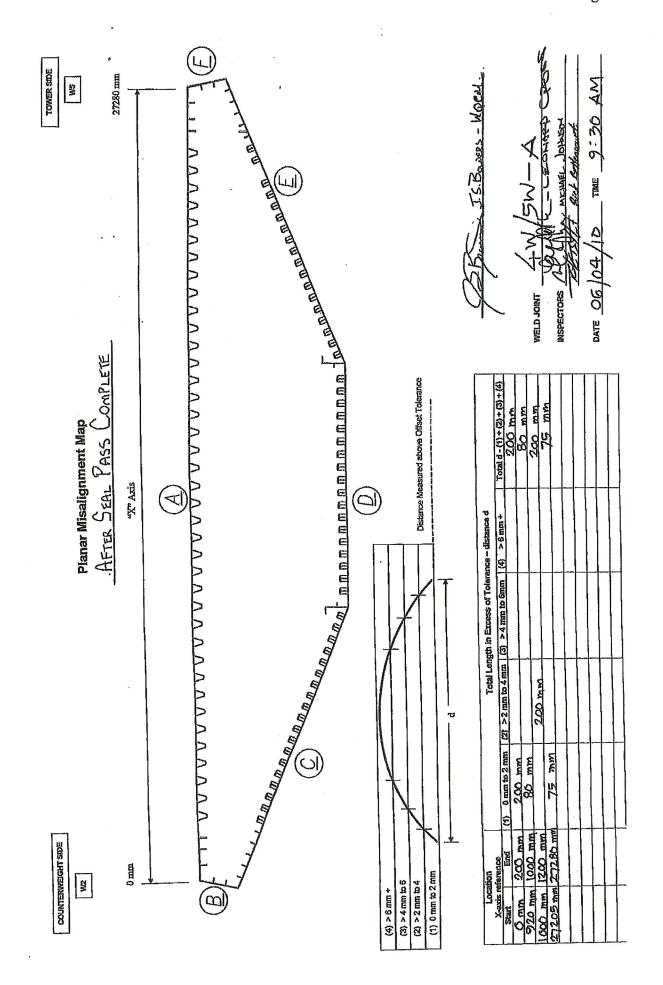
OFFSET MAPS



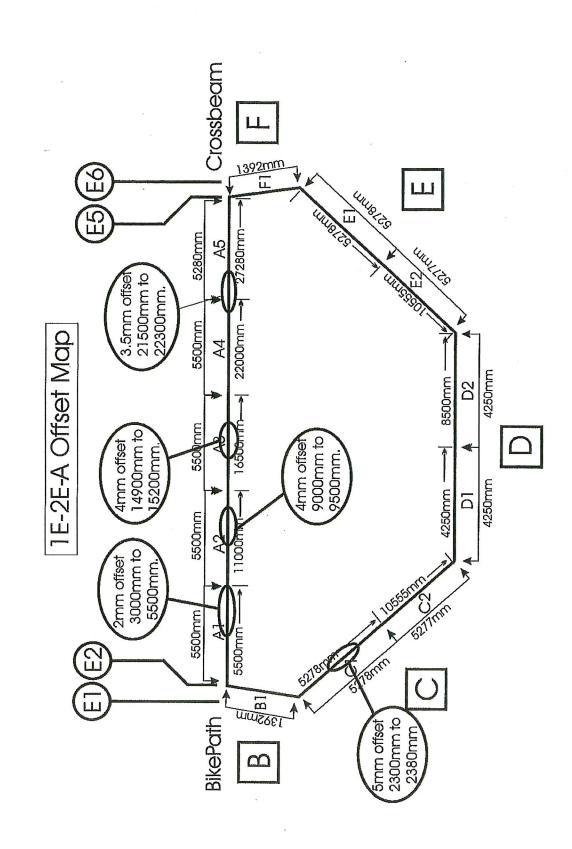


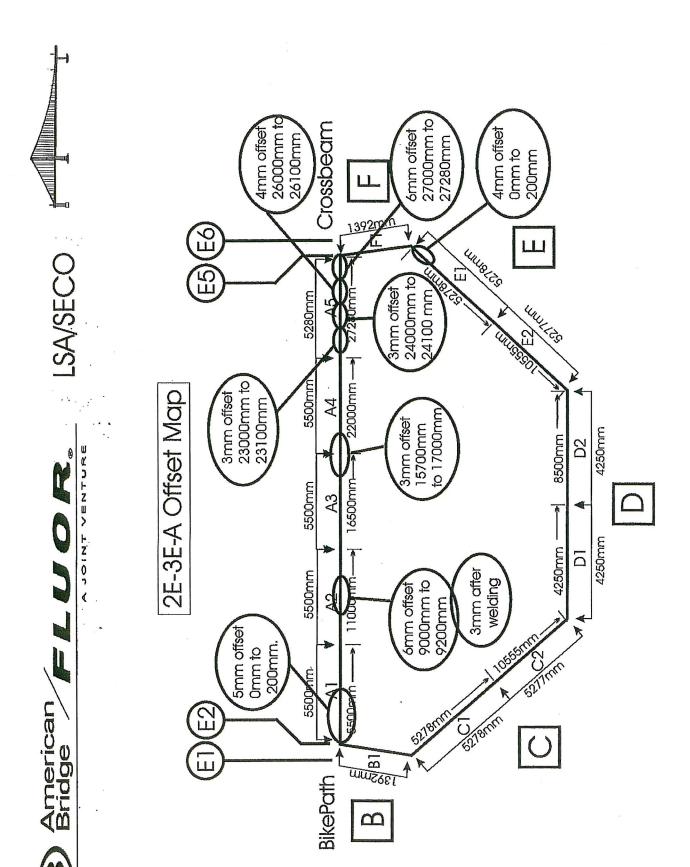


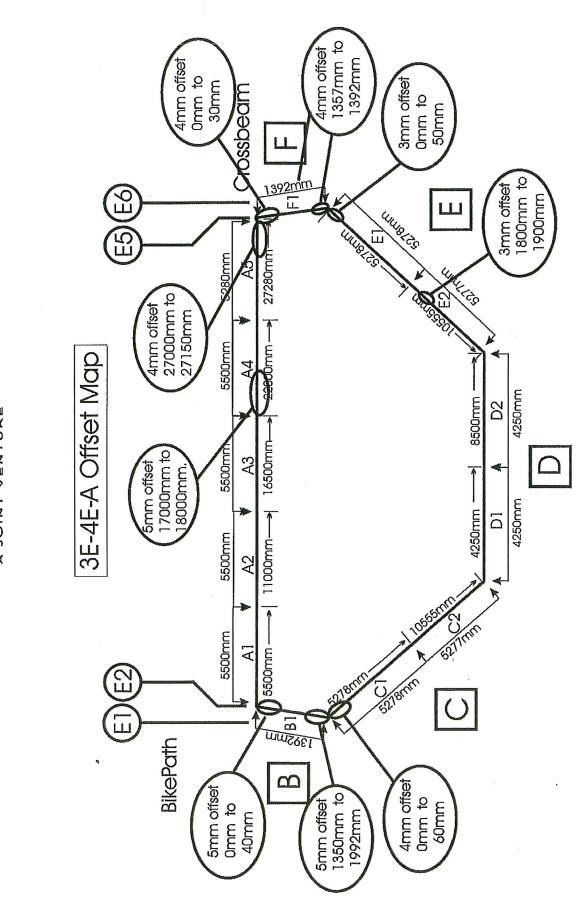










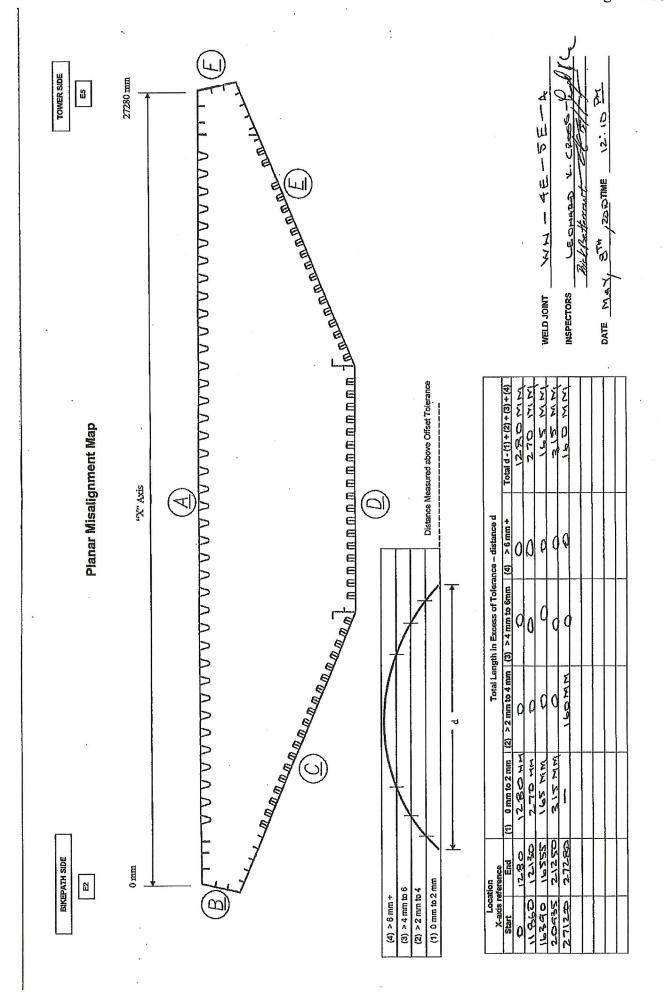


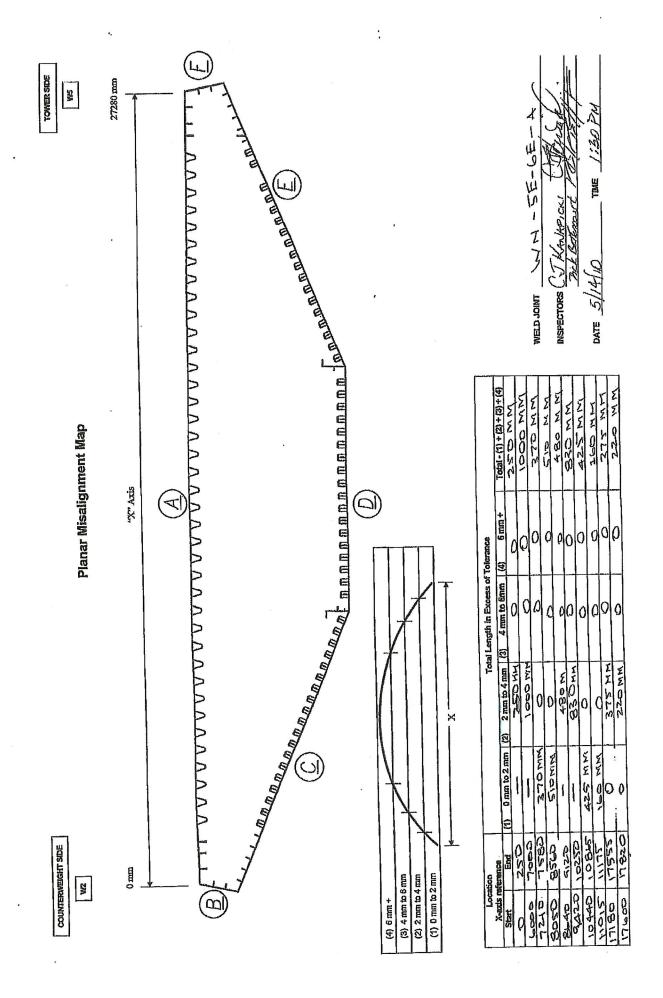


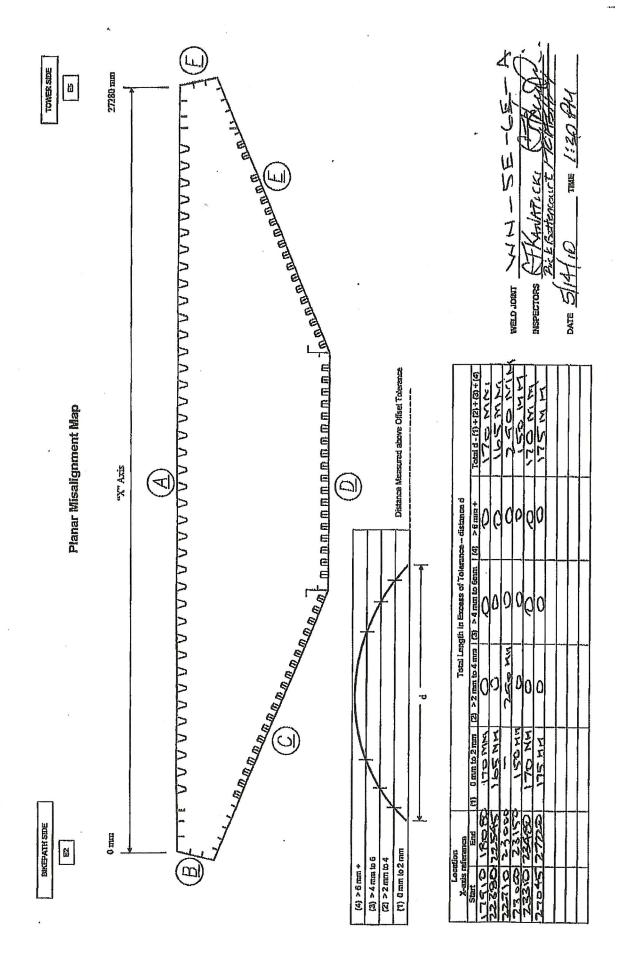
LSA/SECO

American FLUO

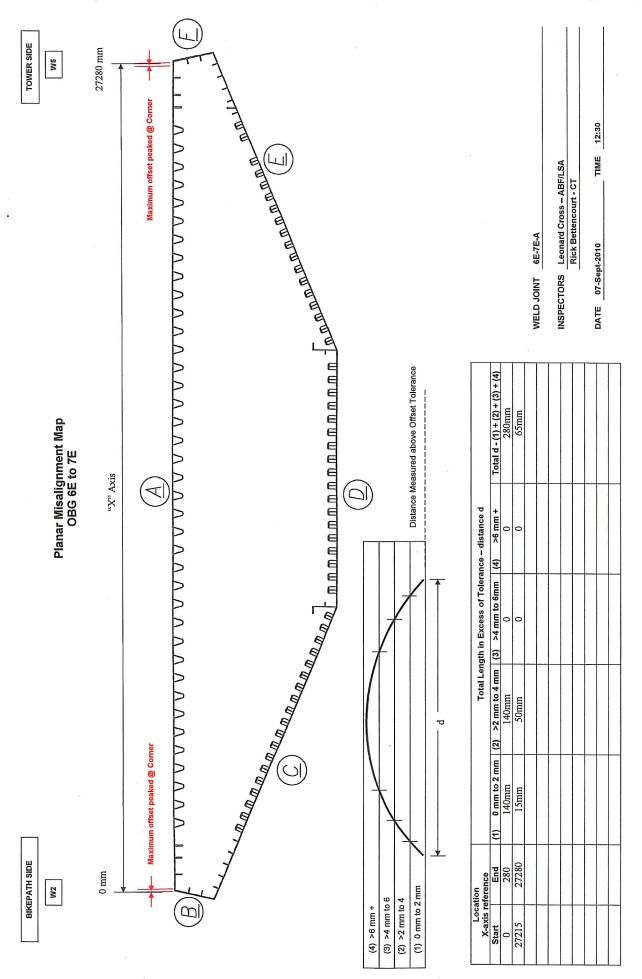








ige 40 of 5



APPENDIX D

DECK PLATE SECONDARY STRESS CALCULATIONS



SAN FRANCISCO OAKLAND BAY BRIDGE **EAST SPAN SEISMIC SAFETY PROJECT SELF-ANCHORED SUSPENSION BRIDGE** (SUPERSTRUCTURE AND TOWER)

Project No.: 660110 Designed By: Kevin Smith Date: 10/29/2010 Page: 1 of 4 Revision: Kilometer Post 13.2 / 13.9

Contract No. 04-0120F4

Bridge No. 34-0006L/R

District 04 County SF Route 80

BOX GIRDER DECK PLATE OFFSET SECONDARY STRESS CALCULATIONS

PURPOSE:

Determine the secondary flexural stress in the OBG deck plate transverse splices due to planar misalignments in the deck plate.

DESIGN APPROACH:

Combine global and local axial stresses in the deck plate due to dead and live load. Detailed calculations are provided below for OBG field splice 1E. Resulting stresses for all of the QBG field splices are provided in the attached table.

DESIGN STANDARD:

AASHTO LRFD Bridge Design Specifications, Second Edition, 1998, Section 6.6.1.2 for Load-Induced Fatigue. Self-Anchored Suspension Bridge Design Criteria, dated 04/08/02. Caltrans Bridge Design Specifications, February 2004.

REFERENCES:

Submittal ABF-SUB-000457R02 and ABF-SUB-001376R00 Contract Plan Sheet Nos. 643, 738 and 741.

CALCULATIONS:

Dead Load Compression in Box Girder at Field Splice 1E:

$$P_{DL} := -200MN$$
 $P_{DL} = -200000 \cdot kN$ (Per Drawing 738)
$$A_{gross} := 2.34351m^2$$
 $A_{gross} = 2.344 \cdot m^2$ (Per Submittal 457R02)
$$f_{DLC} := \frac{P_{DL}}{A_{gross}}$$
 $f_{DLC} = -85.34 \cdot MPa$

Dead Load Moment in Box Girder at Field Splice 1E:

$$M_{DL} := -55.801 MN \cdot m$$
 (Per Submittal 1376R00)
 $c_{y.1E} := 2.365 m$ (Per Submittal 457R02)
 $I_{x.gross.1E} := 11.785 \cdot m^4$ (Per Submittal 457R02)
 $f_{DLM} := \frac{M_{DL} \cdot c_{y.1E}}{I_{x.gross.1E}}$ $f_{DLM} = -11.2 \cdot MPa$

Live Load Compression in Box Girder at Field Splice 1E:

$$P_{I.I.} := -20MN$$

$$P_{DL} = -200000 \cdot kN$$

(Per WDC Discussions)

$$f_{LLC} \coloneqq \frac{P_{LL}}{A_{gross}}$$

$$f_{LLC} = -8.53 \cdot MPa$$

Minimum Live Load Moment in Box Girder at Field Splice 1E:

$$M_{I.I.\ min} := -97.142MN \cdot m$$

$$M_{LL.min} = -97.14 \cdot MN \cdot m$$
 (Per WDC Discussions)

$$f_{LLM.min} \coloneqq \frac{M_{LL.min} \cdot c_{y.1E}}{I_{x.gross.1E}}$$

$$f_{LLM.min} = -19.49 \cdot MPa$$

Maximum Live Load Moment in Box Girder at Field Splice 1E:

$$M_{LL.max} := 85MN \cdot m$$

$$M_{LL,max} = 85 \cdot MN \cdot m$$
 (Per WDC Discussions)

$$f_{LLM.max} \coloneqq \frac{M_{LL.max} \cdot c_{y.1E}}{I_{x.gross.1E}}$$

$$f_{LLM.max} = 17.06 \cdot MPa$$

Minimum Local Live Load Moment in Deck Plate:

$$P_{wheel} := -71kN$$

$$P_{wheel} = -71 \cdot kN$$

(Per SAS Design Criteria

Dated 04/08/02)

IM := 30%

(Per SAS Design Criteria Dated 04/08/02)

a := 312.5 mm

$$e := 287.5 mm$$

$$width_{eff} = 0.78 \cdot m$$

(Per AASHTO LRFD, 1998)

$$M_{LLL.min} := -28.447kN \cdot m$$

 $width_{eff} := 1.3(a + e)$

(Per Service Load Influence Lines)

$$c_{rih} := 83.1mm$$

$$c_{rib} = 0.083 \cdot m$$

$$I_{rib} := .00032534 \cdot m^4$$

$$I_{rib} = 0.0003253 \cdot m^4$$

$$f_{LLL.min} := \frac{(I + IM) \cdot M_{LLL.min} \cdot c_{rib}}{I_{rib}}$$

$$f_{LLL.min} = -9.45 \cdot MPa$$

$$f_{LLL.min.vonMises} := -41MPa$$

Total local live load stress calculated from FEA analysis. Includes 30% impact factor.

Maximum Local Live Load Moment in Deck Plate:

$$M_{LLL.max} := 11.527kN \cdot m$$

(Per Service Load Influence Lines)

$$f_{LLL.max} := \frac{\left[(1 + IM) \cdot M_{LLL.max} \cdot c_{rib} \right]}{I_{rib}}$$

$$f_{LLL.max} = 3.83 \cdot MPa$$

 $f_{LLL.max.vonMises} := 29MPa$

Total local live load stress calculated from FEA analysis. Includes 30% impact factor.

Total Service Load Deck Plate Axial Stress:

$$f_{sum.max} := f_{DLC} + f_{DLM} + f_{LLC} + f_{LLM.max} + f_{LLL.max.vonMises} = -59.02 \cdot MPa$$

$$f_{sum.min} := f_{DLC} + f_{DLM} + f_{LLC} + f_{LLM.min} + f_{LLL.min.vonMises} = -165.57 \cdot MPa$$

Evaluate a 100mm Unit Width of Deck Plate at Splice 1E:

$$w_{100} = 100 mm$$

$$t_{DP} = 20mm$$

$$f_{sec.min} := f_{DLC} + f_{DLM} + f_{LLC} + f_{LLM.min} + f_{LLL.min} = -134.01 \cdot MPa$$

$$P_{100.min} := f_{sec.min} \cdot w_{100}^t DP$$

$$P_{100,min} = -268.03 \cdot kN$$

Calculate Secondary Stresses Due to the Axial Demand and the Plate Misalignment at Splice 1E:

ecc = 7mm

$$M_{ecc.min} := \frac{P_{100.min} \cdot ecc}{2}$$

$$M_{ecc\ min} = -0.938 \cdot kN \cdot m$$

$$Z_{DP} := \frac{w_{100} t_{DP}^2}{4}$$

$$Z_{DP} = 10 \cdot cm^3$$

$$f_{secondary} \coloneqq \frac{M_{ecc.min}}{Z_{DP}}$$

$$f_{secondary} = -93.81 \cdot MPa$$

Plastic Section Modulus Used for Weak Axis Secondary Flexural Stresses

$$f_{total.min} := f_{sum.min} + f_{secondary}$$

$$f_{total.min} = -259.38 \cdot MPa$$

Calculate Tensile Fatigue Stress from Fatigue Truck Load:

$$LF := 0.75$$
 $IM := 15$

$$M_{fatigue} \coloneqq LF \cdot (27.114kN \cdot m) \cdot \left(1 + \frac{IM}{100}\right)$$
 $M_{fatigue} = 23.39 \cdot kN \cdot m$ (Per Fatigue Load Influence Lines)

$$c_{rib} = 0.083 \cdot m$$

$$I_{rih} = 0.0003253 \cdot m^4$$

$$f_{fatigue} := \frac{M_{fatigue} \cdot c_{rib}}{I_{rib}}$$

$$f_{fatigue} = 5.97 \cdot MPa$$

Fatigue Stress in Tension

$$f_{fatigue.vonMises} := 19.24MPa$$

Total local live load fatigue stress calculated from FEA analysis. Includes 15% impact factor and 0.75 load factor.

Calculate Secondary Stresses Due to the Fatigue Truck Load:

$$w_{100} := 100 mm$$

$$t_{DP.fatigue} := 20mm$$

$$P_{100.fatigue} := f_{fatigue} \cdot \text{W}_{100}^t DP.fatigue$$

$$P_{100.fatigue} = 11.95 \cdot kN$$

$$ecc := 7mm$$

$$M_{ecc.fatigue} \coloneqq \frac{P_{100.fatigue} \cdot ecc}{2}$$

$$M_{ecc.fatigue} = 0.042 \cdot kN \cdot m$$

$$f_{secondary.fatigue} \coloneqq \frac{\textit{M}_{ecc.fatigue}}{\textit{Z}_{DP}}$$

$$f_{secondary.fatigue} = 4.18 \cdot MPa$$

$$f_{total.fatigue} := f_{fatigue.vonMises} + f_{secondary.fatigue}$$

$$f_{total.fatigue} = 23.42 \cdot MPa$$

$$\Delta F_{th} := 110 MPa$$

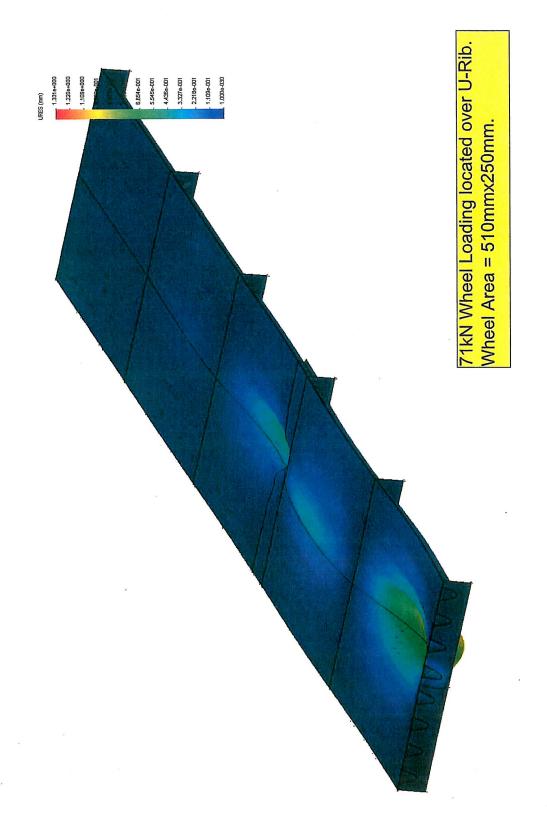
For Detail Category B

$$\Delta F_n := \frac{\Delta F_{th}}{2}$$

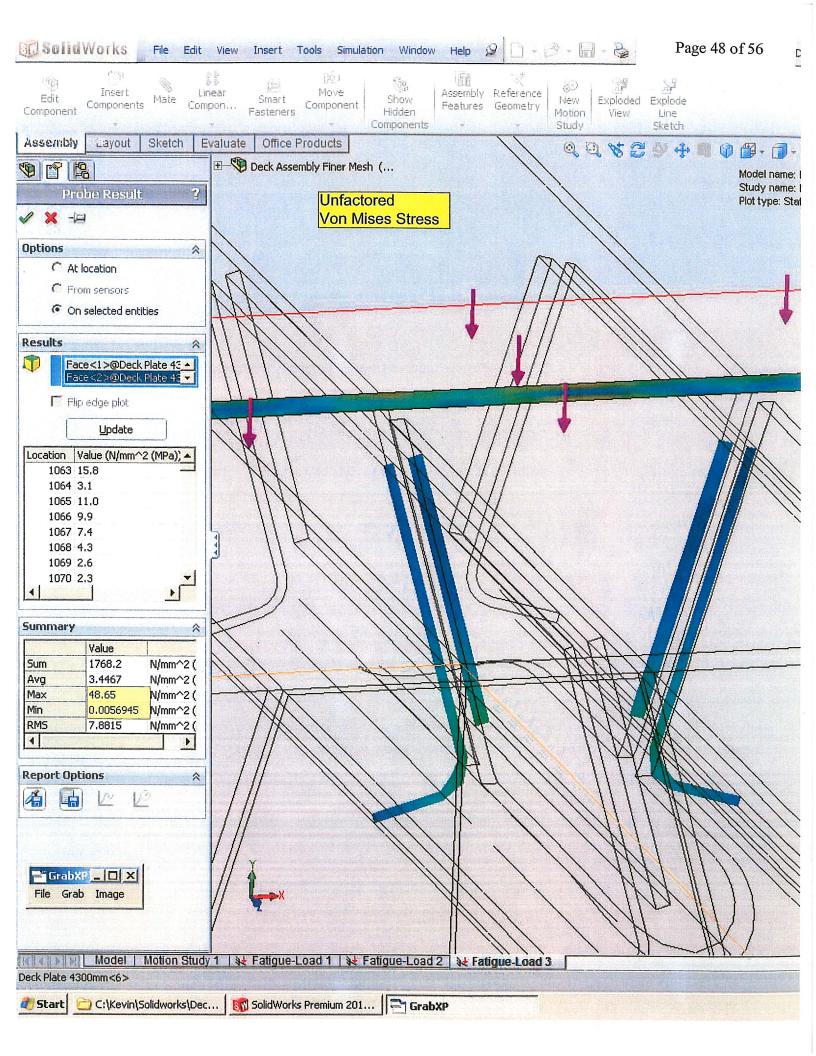
$$\Delta F_n = 55 \cdot MPa$$

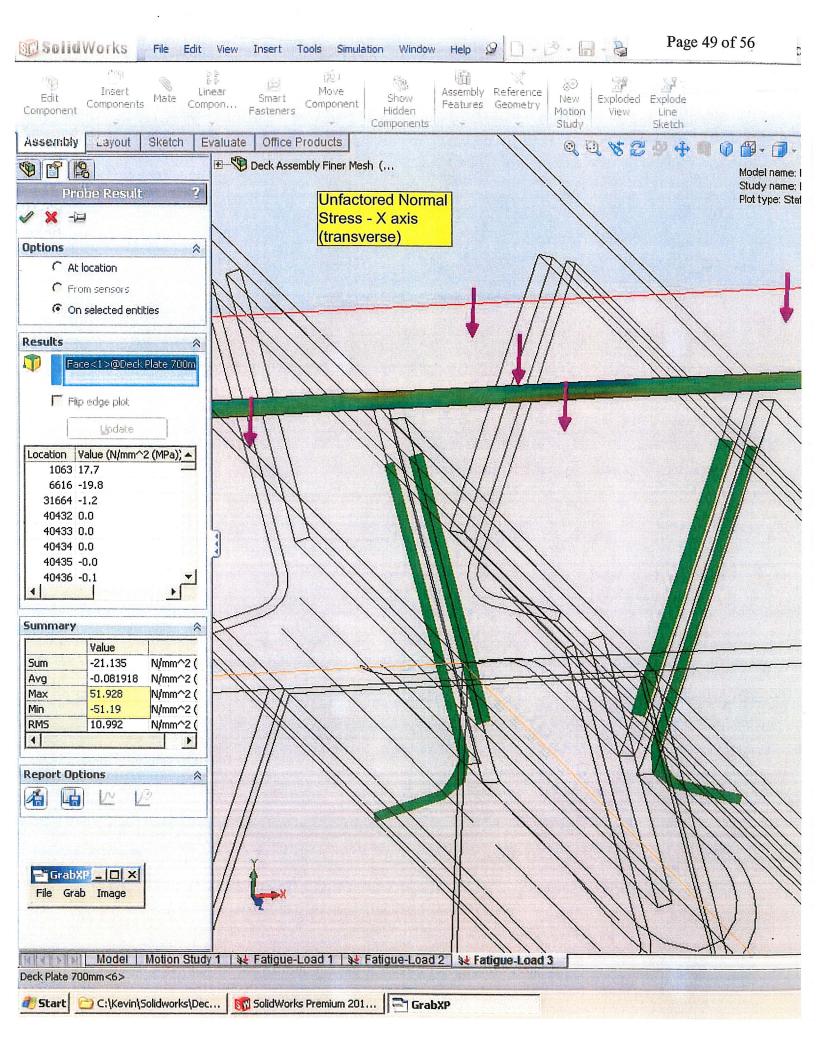
$$DC \coloneqq \textit{if}\left(f_{total.fatigue} < \frac{\left|f_{DLC} + f_{DLM}\right|}{2}, \text{"Fatigue Need Not Be Considered"}, \frac{\left|f_{total.fatigue}\right|}{\Delta F_n}\right)$$

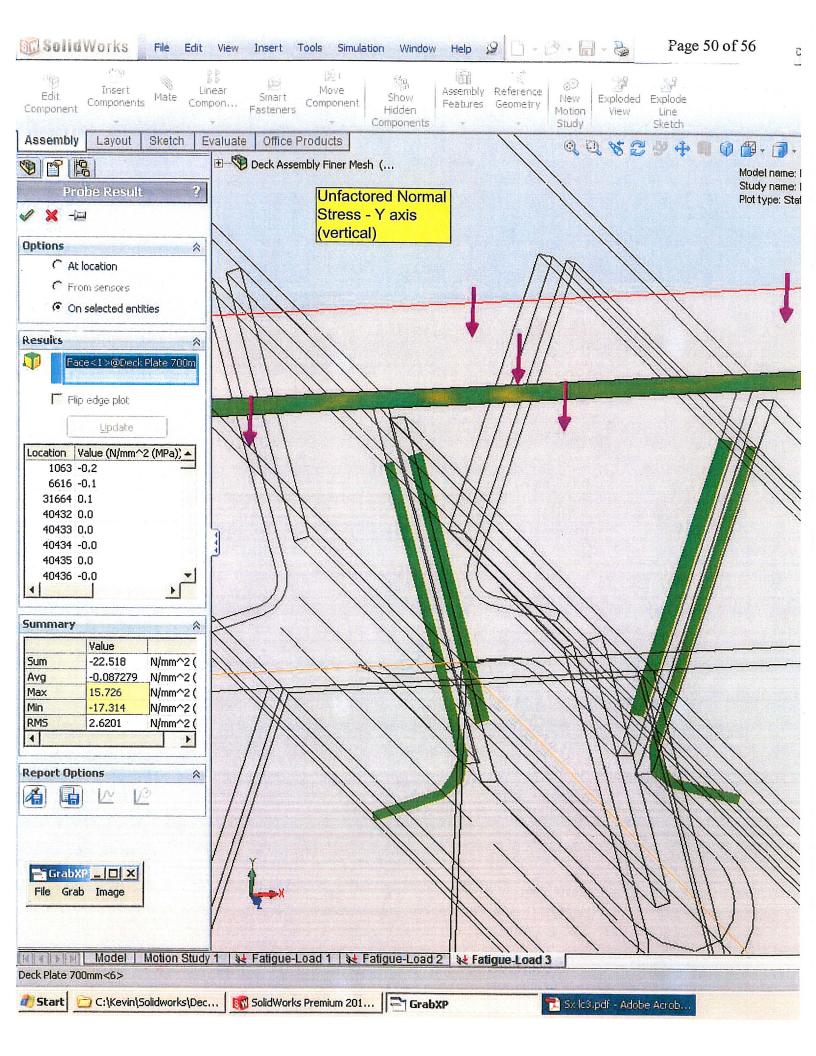
DC = "Fatigue Need Not Be Considered"

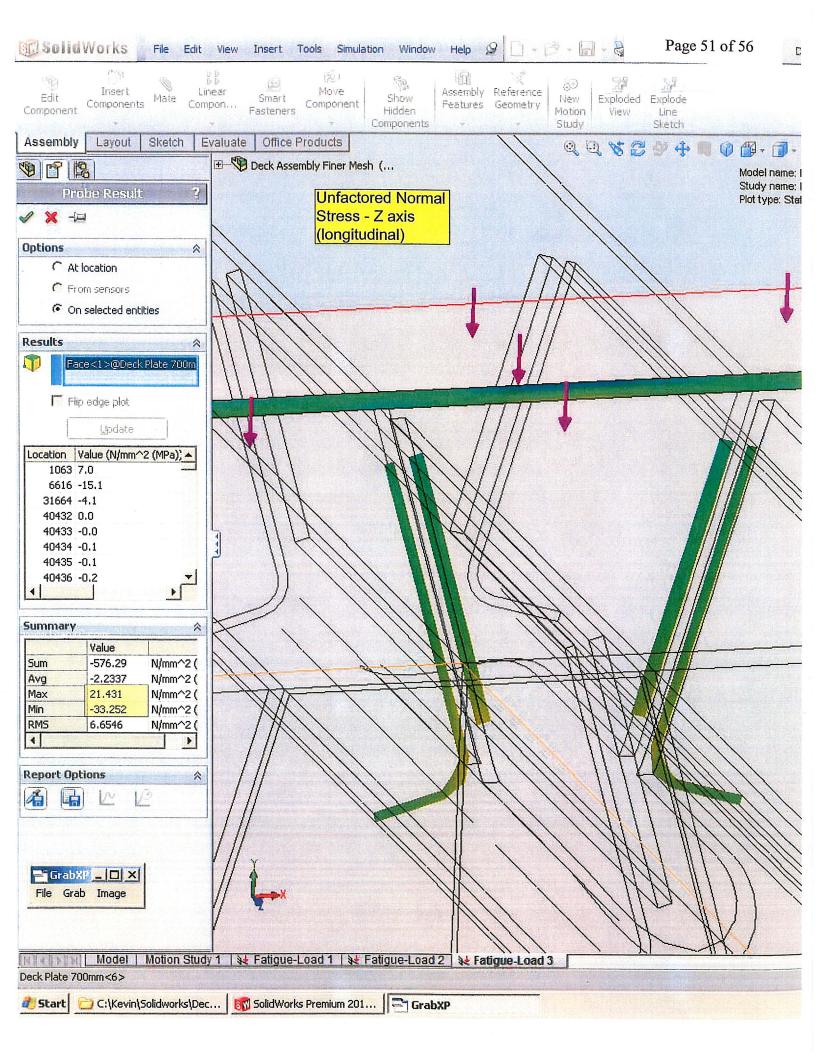








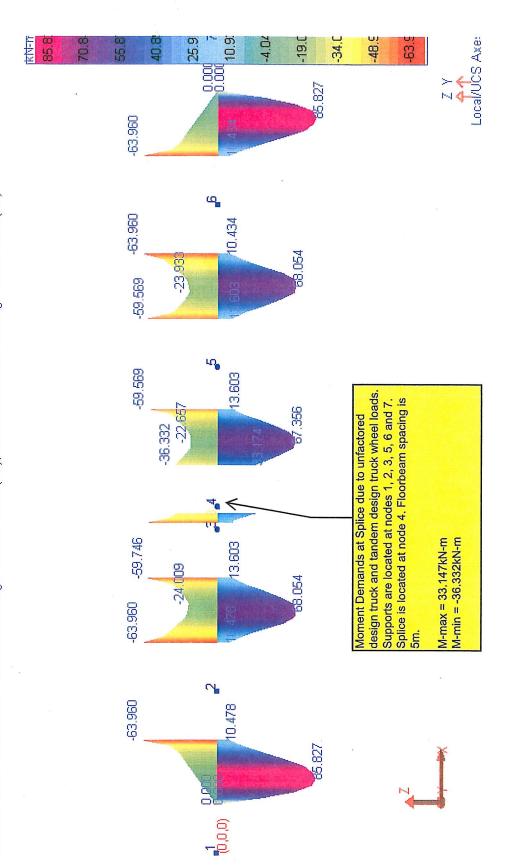




Page 1

Graphics View 1

Member Forces - Moment Mz - Lane/HL-93 Design Tandem Wheel (EE), Lane/HL-93/HS20-44 Design Truck Wheel (EE) Zoom 1.953X

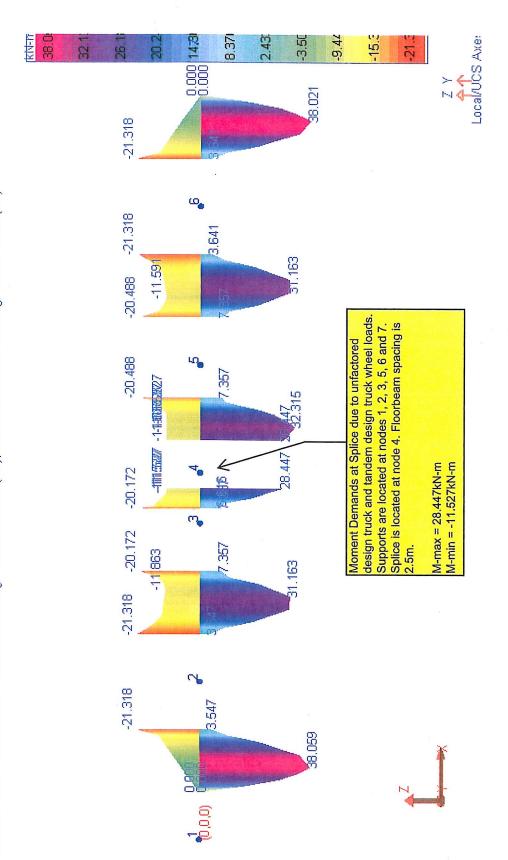


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Last Analysis Run : 9/30/2010 7:54:46 AM

LARSA 4D

Graphics View 1

Member Forces - Moment Mz - Lane/HL-93 Design Tandem Wheel (EE), Lane/HL-93/HS20-44 Design Truck Wheel (EE) Zoom 3.815X



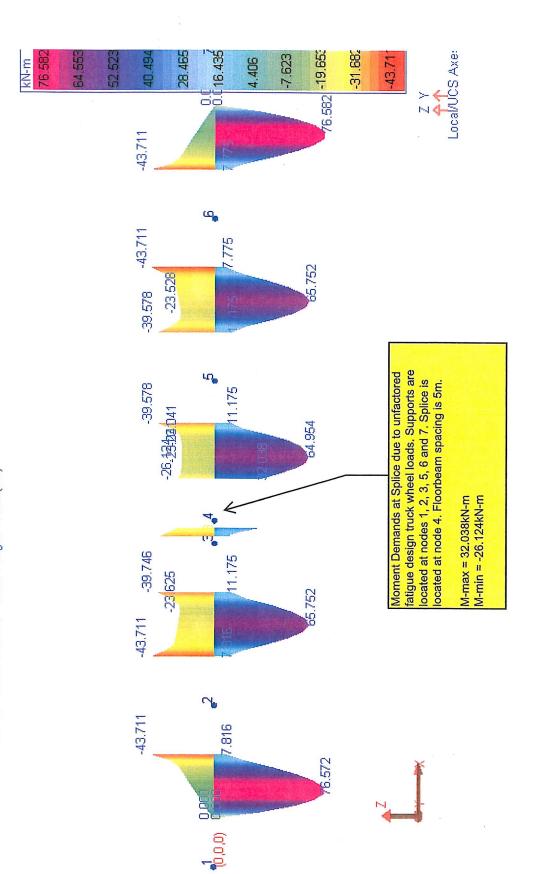
Page 1

American Bridge / Fluor A Joint Venture C:\Documents and Settings\ksmith\Desktop\0BG Planar Misalignment\Deck Plate Influence Lines (2.5m Span).lar Last Analysis Run : 9/30/2010 8:07:34 AM

Page 1

Graphics View 1

Zoom 1.953X Member Forces - Moment Mz - Lane/AASHTO HS-20 Fatigue Truck (EE)



LARSA 4D

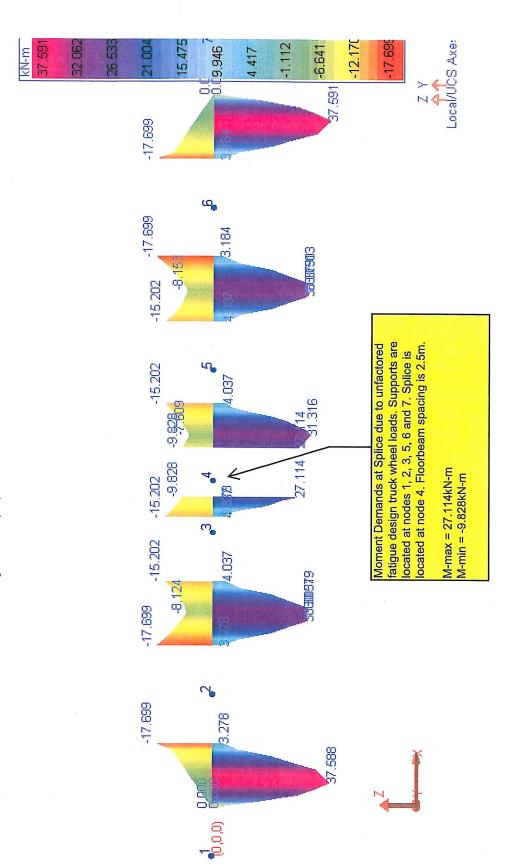
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Last Analysis Run : 9/30/2010 8:07:34 AM

Graphics View 1

DRAFT

Zoom 1.953X Member Forces - Moment Mz - Lane/AASHTO Fatigue Truck (EE)



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Note: Tensile Fatigue Stresses have	been reduced by a factor of 0,837 to	900	wheel load due to the 50mm Epoxy

																									١	TOT	1	43.0	40.4	40.4	40.4	40.4	40,4	40.4	40.4	22.B	22.8	22.8;	40.4	40.4	40.46	40.48	40.4	40.4	40.4	40.4	22.8	46.00
																										TOTAL	-1014	-52.43	-49.81	-49.81	-49.81	49.81	-49,81	-49.81	-49.81	-49.61	-30.79	-30.79	-49.81	-49.81	-49.81	-49.81	-49.81	49.81	-49.81	-49.81	-30.79	20.10
+M (kN-m)	9.828	26.124	26.124	26.124	26.124	20.124	28.124	26.124	26.124	26.124	9.828	9.828	9.628	20.124	28 124	26 124	26.124	26.124	-26.124	26.124	26 124	9 878	9.828		resses	Live Load	Secondary	10.47	7.85	7.85	7.85	7.85	7,85	7.85	7.85	3.58	3.58	3.58	7.85	7.85	7.85	7.85	7.85	7.85	7.85	7.85	3.58	00.0
-M (kN-m) +	4-	-32.038	-32.038	-32.038	-32.038	32,030	-32 038	-32.038	-32.038		-27.114	-27.114	-27.114	32.038	-32.038	32.036	-32.038	-32.038	-32.038	-32.038	32,038	27 114	-27.114		ΞЦ.			32.60	32,60	32.60	32.60	32.60	32.60	32.60	32.60	19.24	19.24	19.24	32.60	32.60	32.60	32.60	32.60	32.60	32.60	32,60	19.24	15.24
+M (kN-m)		36,332	36.332	36.332	36.332	36.332	36.332	36,332	36,332	36,332	11.527	11.527	11.527	36.332	30,332	26,335	36.332	36,332	36.332	36.332	30.332	11 577	11.527			Local Live Loads	TET (Wha)	-41.96	-41,96	-41.96	-41.96	41,96	-41.96	-41.96	41.96	-27.70	-27.20	-27.20	-41.98	41.96	-41.96	41.96	-41,96	41.96	-41.96	-41.96	-27.20	121.20
-M (kN-m)		-33.147	-33.147	-33.147	-33.147	23.14/	33 147	-33.147	-33.147	33.147	-28.447	-28.447	-28.447	-33.147	23 147	23 4 47	-33.147	-33.147	-33.147	-33.147	-33.14/	28 447	-28.447			-	Load Factor	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.73
Fatigue Load	15%	15%	15%	15%	15%	200	144	15%	15%	15%	15%	15%	15%	15%	10%	15.04	15%	15%	15%	15%	1576	785	15%				+10IAL	126.34	101.33	112.56	101.44	90.44	102.48	106.89	116.57	94.20 68.64	40.02	22.53	81.18	112.35	101,51	90.58	107.21	106.95	116.76	94.02	68.31	38.77
Service Load	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30.28	30%	30%	30%	30%	30%	2007	30%		EE		-TOTAL	-259.36	-352.69	-347.77	-314.94	-298.54	-357.82	-378.72	-384.84	-269.34 -94.38	-105.16	-241.42	-342.84	-354.92	-314.00	-296.76	-343.01	75.75°	-382.42	-271.78	-95.72	-110./8
M (MN-m)	985	46	92	120	114	982	44	47	78	108	177	92	82	46	200	120	835	11	44	47	8/9	3 2	92		Combined Stresses	Secondary	Stress (MPa)	180.43	140.18	137.91	122.76	115.19	142.55	152.20	155.02	101.72	27.60	78.70	135.64	139.20	122.33	114,37	135.72	151.85	153.91	102.84	24.06	29.71
Global Live Loads	-97	-116	-162	-170	-123	-102	-192	-232	-242	-85	-53	-19	-97	-116	-162	0/1-	-102	-192	-211	-232	-242	Ş	-19		Con	el.	- 1	-54.10	-38.85	-25.35	-21.32	-24.75	-40.07	-45.30	-38.45	-7.51 45.08	12.41	-56.17	-54.46	-26.85	-20.82	-23.79	-28.51	40.15	-37.15	-8.82	44.24	3.05
Glo Glo	-20	-20	-20	-20	-20	-20	-20	-20	-20	-20	-20	-20	-20	-20	-20	-20	200	-20	-20	-20	-20	07-	-20			Min. (MPa)	-SUM	-705.57	-212.51	-209.86	-192.18	-183.35	-215 27	-226.52	-229.82	-167.63	-77.56	-162.72	-207.20	-213.71	-191.67	-182.39	-207.29	-215.35	-228.51	-168.94	-71.65	-81.07
M (MN-m)	-55.801	26.852	49.512	68.348	99.048	113.784	108.096	68.987	65.882	160,47	355.527	300,082	-41.603	25,396	44.675	62.326	117.65	107.07	92.999	70.567	71.141	133, 130	279.14				Component	3.83	15,66	15.66	15.66	15.66	15.66	15.66	15.66	15.66	3.83	3.83	15.66	15.66	15,66	15.66	15.66	15,66	15.66	15.66	3.83	3.83
Dond Load	-200	-200	-200	-200	-200	-200	-200	2002-	-200	-200	-200	-200	-200	-200	-200	-200	200	-200	-200	-200	-200	202-	2002-			Loads	₹	-14.29	-14.29	-14.29	-14.29	-14.29	-14.29	-14.29	-14.29	-14.29	-9.45	-9.45	-14.29	-14.29	-14.29	-14.29	-14.29	-14.29	-14.29	-14.29	-9.45	-9.45
Cu(m)	0.0831	0.0956	0.0956	0.0956	0.0956	0.0956	0.0956	0.0956	0.0956	0.0956	0.0831	0.0831	0.0831	0.0956	0.0956	0.0956	0.0356	0.0956	0.0956	0.0956	0.0956	0.0936	0.0831			Local Live Loads	+LLL (MPa)	49 14	49.14	49.14	49.14	49.14	49.14	49.14	49.14	49.14	29	29	49.14	49.14	49.14	49.14	49.14	49.14	49.14	49.14	29	28
Local	1 257F.04	2.883E-04	2.883E-04	2.883E-04	2.883E-04	2.883E-04	2.883E-04	2 AR3F-04	2.883E-04	2.883E-04	3.253E-04	3.253E-04	3.253E-04	2.883E-04	2.883E-04	2.883E-04	2.8835-04	2.883E-04	2.883E-04	2.883E-04	2.883E-04	Z.583E-04	3.253E-04				<u>.</u>	41.00	-63.25	-63.25	-63.25	-63.25	-63.25	-63.25	-63.25	-63.25	41.00	-41.00	-63.25	-63.25	-83.25	-63.25	-63.25	-63.25	-63.25	-63.25	41.00	41.00
Cant Di (m)		+	0.014	0.014	0.014	0.014	0.014	0.014		0.014	0.020	0.020	0.020	0.014	0.014	0.014	0.014	0.014	0.014	0.014	0.014	0.014	0.020		Stresses			17.06	21.03	29.84	28.09	21.02	18.92	11.54	19.17	26.71	16.49	17.06	11.42	21.03	28,09	21.02	18.92	10.76	19.17	26.71	35.26	16.63
17.70	7	2.129	2.129	2.129	2.090	2.090	2.090	2.090	2.090	2.090	2.554	2.597	2.365	2.129	2.129	2.129	2.090	2.090	2.090	2.090	2.090	2.090	2.606			Global Live Loads	LLC (MPa) -LLM (MPa) +LLM (MPa)	-19.49	-40.24	-42.28	-30.38	-25.19	-47.48	-57.29	-59.82	-21.02	-3.48	-19.49	-28.94	-40.24	-30,38	-25.19	-47.48	-52.06	-59.82	-21.02	-10.64	-3.51
ŀ	1X (Mr4)	8.561	8,561	8.561	8.451	8.451	8.451	8.451	8,451	8.451	12,798	14.521	11.785	8.561	8.561	8.561	6.451	8.451	8.451	8.451	8.451	6.451	14.455			Glot	LC (MPa) -L	-8.53	-11.03	-11.03	-11.19	-11.19	11.19	-11.19	-11.19	-11.19	-7,89	-8.53	-11.03	-11.03	-11.19	-11.19	-11.19	11.19	-11.19	Н	-8.19	┨
1000	7 34351	1.8132	1.8132	1.8132	1,78795	1.78795	1.78795	1.78705	1.78795	1,78795	2.44097	2,53607	2.34351	1,8132	1.8132	1.8132	1.78795	1.78795	1,78795	1,78795	1.78795	1.78795	2.53189			oad	DLM (MPa) I	-11.20	12.31	17.00	24.50	28.14	26.73	17.06	16.29	39.69	53.67	-8.35	6.32	11.11	25.00	29.10	26.48	23.00	17.59	38.37	70.12	50,32
Δı	(E) 0	0.008	0.006	900'0	9000	900'0	9000	9000	0.006	9000	0.006	900'0	9000	900'0	0.008	0.006	9000	0.006	9000	900'0	9000	9000	9000	2000		Dead Load	DLC (MPa) DLM (MPa)	-85.34	-110.30	-110.30	-111.86	-111.86	-111.86	-111.86	-111.86	-111.86	-78.86	-85.34	-110.30	-110.30	-111.86	-111.86	-111.86	-111.86	-111.86	-111.86	-81.93	-78.99
Panel	i co	100	24	59	37	47	6	20	3 58	109	117.5	124.5	13	19	24	53	34	2 2	11	92	8	8	1745	2	٦	Panel	Point	5 5	24	59	37	47	19 14	88	92	109	124.5	13	19	20 24	37	47	61	7.	8 %	109	117.5	124.5
Splice		3 2	38	4E	5E	39	7	3 10	10E	11	12E	13E	JW.	ZW	38	₩.	NG G	3	M	П	1	╅	ME!	1		Splice		1 1	1 14	4€	5E	9E	3/1	38	10E	116	13E	W	ZW.	WE S	MG.	M9	2	88	10W	-	12W 1	-
		_	_	_	_	31	117-	3	_	_	_	٢	۲	_	_	_		NI"	-N	_	_		_	1	•	_	٦	_	_	_	_	NE	17-3	1	_		_		_	_	_	II.	IIT:	M	_	_	_	_

	D/C	N/A	76.0	76.0	N/A	76.0	0.97																				
	Fatigue Check Req'd	ON	ON	ON	NO	ON	ON	NO	O.	NO	NO	NO	YES	YES	ON	ON	NO	ON	NO	NO	NO	NO	NO	ON	ON	YES	YES
	Allowable Fatigue Stress	25.00	34.50	34.50	34.50	34.50	34.50	34.50	34.50	34.50	34.50	34.50	55.00	55.00	55.00	34.50	34.50	34.50	34.50	34.50	34.50	34.50	34.50	34.50	34.50	55.00	92.00
Fatigue Stress	Fatigue Category	8	o	0	o	၁	ن	ပ	c	o	ပ	o	В	В	0	٥	o	၁	O	υ	2	0	υ	0	0	8	8
	2 x Fatigue < DL	YES	ON	NO	YES	ON	ON																				
מופסה סוופסים	DL Compression	96,54	103,62	97.99	93.31	87.36	83.72	85.13	88.78	94.60	95.57	72.17	10.98	25.19	93.69	103.99	99.19	94.80	86.86	82.76	85.38	88.86	94.41	94.27	73.49	11.82	28.67
	Tonsile Faligue Stress	23.42	43.07	40.46	40.46	40.46	40.46	40.46	40.46	40.46	40.46	33.86	22.82	22.82	22.82	40.46	40.46	40.45	40.46	40.46	40.46	40.46	40.46	40.46	33,86	22.82	22.82
	/C (Fy=361)	0.68	1.02	0.93	16.0	0.83	0.78	06.0	0.94	66.0	1.01	0.71	0.25	0.28	0.63	0.90	0.93	0.92	0.82	0.78	06'0	0.94	66'0	1.00	0.71	0.25	0.29
95505	D/C (Fy=345) D/C (Fy=381	0.75	1.12	1.02	1.01	0.91	0.87	66.0	1.04	1.10	1.12	0.78	0.27	0.30	0.70	0.99	1.03	1.02	16.0	0.86	66.0	1.04	1.10	1.11	0.79	0.28	0.32
Service Load Stresses	Combined Stress		387.27	352.69	347.77	314,94	298.54	342.54	357.82	378.72	384.84	269.34	94.38	105.16	241.42	342.84	354.92	350.55	314.00	296.76	343.01	357.97	378.00	382.42	271.78	95.72	110.78
	Yield Stress	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345.00	345,00	345.00	345.00	345.00	345.00
Panol	Point	13	49	24	59	37	47	61	71	88	88	109	117.5	124.5	13	19	24	53	37	47	9	71	88	98	109	117.5	124.5
Splice	Number	3	2E	3E	4E	3E	96	7.E	96	36	10E	116	12E	13E	¥	200	3W	W4	WG.	Mg	ž	W8	W6	10W	11W	12W	13W

AB) American FLUOI